# Thermal Conductivity of Propane in the Temperature Range 25–305°C and Pressure Range 1–70 MPa

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A coaxial cylinder method was used to measure the thermal conductivity of propane in the pressure range from 1 to about 70 MPa and in the temperature range from room temperature to  $305^{\circ}$ C. The behavior of the thermal conductivity in the critical region was carefully investigated.

KEY WORDS: critical region; high pressure; propane; thermal conductivity.

#### 1. INTRODUCTION

Up to now, the most significant measurements of the thermal conductivity of propane were made by Leng and Comings [1], Ryabtsev and Kazaryan [2], Carmichael et al. [3], Brykov et al. [4], and more recently at lower temperatures by Roder and Nieto de Castro [5]. Tentative tables and correlations representing the thermal conductivity of propane as a function of temperature and pressure (or density) have been reported [6, 7], but it has been pointed out that the set of data used to generate these tables should be improved, especially in the critical region, where almost no data exist. Thus, new measurements of the thermal conductivity of propane were performed over wide temperature and pressure ranges (25–305°C and up to 70 MPa); special attention was paid to the critical region.

These measurements are a part of an extensive program carried on in our laboratory to determine the thermal conductivity of hydrocarbons. The thermal conductivities of methane [8, 9], ethane [9, 10], *n*-butane [11], and isobutane [12] are already available.

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#### 2. EXPERIMENTAL METHOD

The thermal conductivity was measured with a concentric cylinder apparatus described in earlier publications [13, 14]. The experimental procedure was identical to that used, for example, in measuring the thermal conductivity of *n*-butane [11]. The fluid is located in the annular gap between two coaxial cylinders with the axis in the vertical direction. The thermal conductivity was determined by measuring the temperature difference between the inner and the outer cylinders as a function of the energy dissipated from the inner cylinder. The temperature difference between the two cylinders varies from  $0.3^{\circ}$ C close to the critical point to about 2°C far away from the critical point; this temperature difference was measured with an accuracy of approximately  $\pm 0.003^{\circ}$ C. The temperature was measured with an accuracy of  $\pm 0.02^{\circ}$ C, and the pressure with an accuracy of 0.1%.

To determine the thermal conductivity, we need to consider the corrections due to heat transferred by radiation, spurious heat flow from the inner to the outer cylinder through the solid centering pins and the wires, heat transfered by convection, and the effects of a possible temperature jump at the boundaries of the fluid layer and surfaces of the cylinders [14].

We calculated the radiation correction from the Stefan–Boltzmann radiation law, assuming that the absorption of radiation by the fluid could be neglected.

The correction for heat losses through the solid parts of the cell was determined from a set of calibration measurements with argon, neon, and helium, for which the thermal conductivity is known with considerable accuracy [15]. These calibrations were performed at pressures of 1 MPa for argon, 5 MPa for neon, and 10 MPa for helium, i.e., at pressures for which any temperature jump can be neglected [14].

The convection which takes place in the cell is assumed to be laminar. In that case, the correction for convection heat flow  $Q_c$  is approximated by the relation [16]

$$Q_{\rm c} = \operatorname{Ra} \frac{2\pi r}{720} \,\lambda \,\Delta T \tag{1}$$

where Ra is the Rayleigh number,  $\lambda$  is the thermal conductivity of the fluid, r is the radius of the inner cylinder, and  $\Delta T$  is the temperature difference between the cylinders.

The measurements were not made at pressures lower than 1 MPa. Under this condition, the temperature-jump effect can be neglected.

#### 3. RESULTS

The experimental data are presented in Table AI (Appendix). The thermal conductivity was determined with a reproducibility of 1% and an estimated accuracy of about 2% except in the critical region, where a reliable estimate of the accuracy is more difficult to make.

The density  $\rho$  was calculated from the equation of state developed by the National Bureau of Standard (Boulder) [17]. In this equation the critical parameters are the following:

$$T_c = 369.85 \text{ K};$$
  $P_c = 4.2471 \text{ MPa};$   $\rho_c = 220.5 \text{ kg} \cdot \text{m}^{-3}$ 

The experimental data are shown in Fig. 1 as a function of the density along quasi-isotherms.

We have restricted the comparison of our data with the tabulated values proposed by Vargaftik et al. [6] and the correlation proposed by Holland et al. [7]. This comparison is shown in Fig. 2 for 0.1- and 30-MPa isobars. The experimental values at 0.1 MPa have been obtained by



Fig. 1. The thermal conductivity of propane as a function of density and temperature.



Fig. 2. Comparison between our experimental values of the thermal conductivity of propane at 0.1 MPa (extrapolated) and 30 MPA and the proposed values in Refs. 6 and 7.

extrapolation. It appears that our experimental values are in better agreement with the tabulated values proposed by Vargaftik et al.

We have presented in Fig. 3 the critical enhancement  $\Delta \lambda_{\rm c}(\rho, T)$  of the thermal conductivity of propane and in Fig. 4 the behavior of the critical enhancement at  $\rho = \rho_{\rm c}$  as a function of the temperature distance from the critical temperature.

$$\Delta\lambda_{\rm c}(\rho, T) = \lambda(\rho, T) - \lambda_{\rm B}(\rho, T) \tag{2}$$

with

$$\lambda_{\rm B}(\rho, T) = \lambda_0(T) + \Delta\lambda(\rho) \tag{3}$$

 $\lambda_0(T)$  is the dilute-gas thermal conductivity and  $\Delta\lambda(\rho)$  is the "normal" density effect, which has been deduced from the behavior of the thermal conductivity far away from the critical point.



Fig. 3. The propane thermal conductivity critical enhancement.



Fig. 4. The critical enhancement of the thermal conductivity of propane for the critical density as a function of the temperature difference  $\Delta T (\Delta T = T - T_c)$ .

 $\lambda_0(T)$  can by represented by

$$\lambda_0(T) = \sum_{i=0}^4 A_i T^i \tag{4}$$

with  $A_0 = 0.0242964$ ,  $A_1 = -3.02775 \times 10^{-4}$ ,  $A_2 = 1.58466 \times 10^{-6}$ ,  $A_3 = -2.66226 \times 10^{-9}$ , and  $A_4 = 1.73834 \times 10^{-12}$ . *T* is in K, and  $\lambda_0$  in W  $\cdot$  m<sup>-1</sup>  $\cdot$  K<sup>-1</sup>.

 $\Delta\lambda(\rho)$  can be written

$$\Delta\lambda(\rho) = \sum_{j=1}^{5} B_j \rho^j$$
(5)

with  $B_1 = 4.13863 \times 10^{-5}$ ,  $B_2 = 2.47025 \times 10^{-7}$ ,  $B_3 = -6.58197 \times 10^{-10}$ ,  $B_4 = 1.29066 \times 10^{-12}$ , and  $B_5 = -4.25454 \times 10^{-17}$ .  $\rho$  is in kg  $\cdot$  m<sup>-3</sup>, and  $\lambda$  in W  $\cdot$  m<sup>-1</sup>  $\cdot$  K<sup>-1</sup>.

Figure 3 shows that the maximum of  $\Delta \lambda_c(\rho_c)_T$  occurs at  $\rho = \rho_c$  except for the 373.8 K isotherm, for which the maximum is located at  $\rho > \rho_c$ . Error bars on *P*, *T*, and eventually the equation of state (nonscaled equation od state in the critical region) can explain this shift. A small systematic error on *P* or *T* along this quasi-isotherm can also be possible.

The critical thermal conductivity excess along the critical isochore can be written [18]

$$\Delta\lambda_{\rm c} = R \frac{k_{\rm B}T}{6\pi\eta\xi} \rho_{\rm c} C_{\rm p}^{\rm c} \exp(-At^2) \tag{6}$$

where R and A are adjustable constants,  $k_{\rm B}$  is the Boltzmann constant,  $\eta$  is the shear viscosity,  $\xi$  is the long-range correlation length,  $C_{\rm P}^{\rm c}$  is the critical part of the specific heat at constant pressure, and  $t = (T - T_{\rm c})/T_{\rm c}$  is the reduced temperature distance from the critical temperature  $T_{\rm c}$ .

Our data can be fitted with the following values of the parameters:

$$R = 1.2$$

$$\xi = \xi_o t^{-\nu} (1 + a_{\varepsilon} t^{\Delta})$$
(7)

with  $\xi_0 = 1.934$  Å,  $a_{\xi} = 1.083$ , v = 0.63, and  $\Delta = 0.5$ ;

$$C_{\rm P}^{\rm c} = \frac{T}{\rho_{\rm c}} \left(\frac{dP}{dT}\right)_{\rho_{\rm c}}^2 K_{\rm T}$$
(8)

with  $(dP/dT)_{\rho_c} = 7.8 \ 10^4 \ \text{Pa} \cdot \text{K}^{-1}$ ; and

$$K_{\rm T} = K_{\rm T}^{\rm o} t^{-\gamma} (1 + a_{\rm x} t^{\rm A}) \tag{9}$$

where  $K_T^{\circ} = 1.1586 \ 10^{-8} \ \text{Pa}^{-1}$ ,  $a_x = 1.68$ , and  $\gamma = 1.24$ .

 $\eta$  is calculated using Eqs. (1), (3), (4), and (7) of Ref. 7.

 $K_T^{\circ}$ ,  $a_x$ ,  $\xi_{\circ}$ , and  $a_{\xi}$  have been estimated from the reduced compressibility and the reduced correlation length analysis proposed by Garrabos [19, 20].

## 4. CONCLUSION

In this paper we have presented new results on the thermal conductivity of propane mainly at temperatures greater than the critical temperature over a wide density range. The present data should help in making data correlation with a greater accuracy, especially in the critical region.

## APPENDIX

<i>Т</i> (К)	P (MPa)	$(kg \cdot m^{-3})$	$\lambda (\mathbf{m}\mathbf{W}\cdot\mathbf{m}^{-1}\cdot\mathbf{K}^{-1})$
302.90	0.83	17.0	21.3
300.5 <sub>8</sub>	0.88	18.4	22.7
300.57	0.91	19.2	22.9
297.8 <sub>0</sub>	1.38	494.4	96.3
297.2 <sub>7</sub>	3.70	501.8	99.2
297.0 <sub>6</sub>	7.25	510.7	104.2
296.7 <sub>1</sub>	7.20	511.1	104.7
298.4 <sub>4</sub>	8.70	512.1	101.8
298.29	12.3	520.2	106.4
298.04	15.5	525.7	108.9
297.8 <sub>5</sub>	18.1	530.2	110.2
296.81	21.2	536.1	115.1
296.47	25.4	542.4	120.2
296.1 <sub>8</sub>	27.9	546.0	122.2
295.9 <sub>5</sub>	27.9	546.2	123.4
298.0 <sub>8</sub>	35.8	553.7	126.9
298.0 <sub>2</sub>	38.8	558.4	127.0
357.3 <sub>8</sub>	4.03	372.3	69.8
356.9 <sub>6</sub>	5.20	390.6	70.8
356.81	6.75	406.2	75.5
357.1 <sub>0</sub>	11.40	434.4	84.8
357.25	21.3	468.8	93.4
356.3 <sub>5</sub>	29.8	489.3	99.7
374.85	3.02	60.2	31.8
374.63	3.415	74.0	33.9

Table AI. Thermal Conductivity of Propane

T	P (MPa)	$\rho$	$\hat{\lambda}$
(K)	(MI a)	(Kg III )	
374.4	3.98	102.1	38.0
374.3	4.03	105.7	40.2
374.27	4.26	126,4	43.0
373.9	4.32	135.9	47.2
373.9	4.445	161.8	55.0
373.8	4.485	178.7	60.1
373.7	4.535	208.3	71.6
373.8	4.55	219.9	77.8
373.8	4.56	225.7	78.6
373.8	4.58	236.8	79.8
373.82	4.59	240.4	79.2
373.8	4 61	246.9	77.6
373.8	4.64	257.8	73.6
373.9	4 73	278.3	67.6
373.9	4.83	291.1	65.6
373.9	5.04	308.8	64.4
373.9	5.24	319.8	64 4
374.0	5.54	331.2	65.0
373.0	5.95	3/3.0	66.3
376.0	3.00	100.1	38.7
375.0	J.775 1 20	100.1	12.0
375.9	4.29	122.7	42.9
375.87	4.545	100.1	53.0
2757	4.055	190.0	66.0
2756	4.045	204.0	68.6
275.6	4.005	204.9	70.0
2756	4.005	213.5	70.9
373.07 275 5	4.735	233.9	70.9 68 A
$373.3_7$	4.775	249.8	08.4 67.6
373.31 275.2	4.03	200.1	65.1
275 1	4.93 5.04	203.0	64.7
$375.1_7$	5.04	297.2	64.7
373.2 <sub>1</sub>	5.55	324.0	04.7
375.09	6.07	257.0	67.5
373.09	0.91	01.0	07.5
380.2	J.705 A 57	71.7	57.0 AA A
380.0	4.525	162.0	44.4 57.8
370.09	4.705	1876	52.0 50 K
379.8	4.20	200.6	57.0
370.8	4.93	200.0	63.3
370.8	4.705	203.7	66.0
370.04	5.04	243.2	65.6
3700	5.13 <sub>5</sub> 5.24	241.0	65.3
380.0	5.245	230.7	63.5 64 7
500.08	5,51	200.0	04.7

Table AI. (Continued)

Т	Р	0	λ
(K)	(MPa)	$(kg \cdot m^{-3})$	$(\mathbf{m}\mathbf{W}\cdot\mathbf{m}^{-1}\cdot\mathbf{K}^{-1})$
380.2	6.06	215.4	
380.2	6.00	313.4	65.1
380.3	0.55 <sub>5</sub>	331.0	66.1
380.3	7.07	342.7	07.0
382.3	2.02	551.0	08.3
382.1	5.75 <sub>5</sub> 4.48	00.2	50.5 42.6
382.0-	4.405	166.4	42.0
383.4	5.03	171 7	53.4
383.4	5.03	1714	54.7
382.0	5.00	181.1	56.5
382.04	5.05	190.5	57.5
382.07	5.15	210.7	62.3
381.8	5.13	210.6	61.7
382.20	5.34	242.8	63.5
382.3	5.56	268.2	64.3
382.47	6.045	301.1	65.0
382.56	7.095	334.8	66.8
382.5,	$7.97_{5}^{-1}$	351.4	68.6
386.7 <sub>0</sub>	4.505	110.0	39.9
386.4 <sub>6</sub>	4.525	111.6	40.3
386.1,	5.225	176.4	53.3
386.15	5.56 <sub>5</sub>	225.5	60.6
386.1 <sub>5</sub>	6.07 <sub>5</sub>	276.0	62.6
386.4	11.90	384.8	73.6
386.17	0.99	14.6	29.3
387.3	0.99	14.6	29.8
386.23	2.52	43.3	31.6
388.6,	2.51	42.6	32.2
388.3	4.02	85.5	37.7
388.09	5.04	144.2	47.1
388.0 <sub>2</sub>	5.56	205.8	58.4
300.03	5,755 5,95	230.0	60.4
388.0	5.825	237.6	60.7
388.0	5.90 <sub>5</sub>	246.0	60.7
388.0	5.92	247.3	61.1
3879	6.05 <sub>5</sub>	259.5	61.6
387.9.	6.24	200.1	63.1
387.9	6 36	274.0	02.0
387.9	6.56	201.1	62.6
387.7	7.09	313.0	64 3
387.8,	8.09	336.8	66 2
387.8,	9.095	352.7	68.2
387.9 <sub>7</sub>	10.33	367.3	70.3

Table AI. (Continued)

<i>T</i> (K)	P (MPa)	$(kg \cdot m^{-3})$	$\lambda$ (mW·m <sup>-1</sup> ·K <sup>-1</sup> )
200.0			
388.89	30.0	455.4	90.7
389.3 <sub>8</sub>	40.0	477.9	98.8
389.23	50.0	495.0	105.1
389.2 <sub>8</sub>	60.0	509.0	111.7
433.9 <sub>1</sub>	0.99	12.7	35.7
433.51	2.51	35.2	37.2
433.57	5.06	85.2	41.1
432.8 <sub>2</sub>	5.03	84.8	41.2
432.6 <sub>5</sub>	7.58	161.4	49.7
431.83	10.11	248.0	59.2
432.5 <sub>1</sub>	10.11	245.8	58.8
432.6 <sub>4</sub>	11.93	285.1	63.0
432.7 <sub>2</sub>	14.40	319.8	67.1
432.29	20.0	366.5	73.8
432.3 <sub>0</sub>	30.0	411.6	85.1
432.3	40.0	440.0	93.0
432.2 <sub>8</sub>	50.0	461.2	99.4
432.25	60,0	478.2	105.6
432.26	70.0	492.4	111.1
432.2	81.2	506.0	116.0
480.67	0.99	11.3	42.5
480.07	2.35	28.3	43.7
479.6	5.05	67.7	46.5
479.2	10.15	164.5	55.4
479.1	12.59	212.4	60.3
476.3	14.50	247.8	64.2
476.0	15.2	258.1	65.3
475.8	20.0	308.9	71.7
475.8	20.1	309.8	71.6
475.8	25.0	343 1	76.9
4757.	30.0	368.0	81.6
476.2	30.0	367.5	80.9
475.6	35.0	387 3	86.0
475.4	40.0	403.4	80.5
4749.	50.2	429.6	97.0
474.6	60.0	440 1	102.5
474 5.	71.9	468 3	102.5
528.0-	0.98	10.1	50.6
527.9	2.50	26.7	51.4
527.7	5.03	57.7	52.7
527.5	5.03	57.2	53.2
528 0-	5.03	57.2	52.0
528.6	10.0	125.2	55.7
520.04	10.10	123.2	50.0

Table AI. (Continued)

Т (К)	P (MPa)	$(\text{kg} \cdot \text{m}^{-3})$	$(\mathbf{m}\mathbf{W}\cdot\mathbf{m}^{-1}\cdot\mathbf{K}^{-1})$
527.3 <sub>3</sub>	12.57	163.1	62.8
529.0 <sub>6</sub>	14.7	190.8	65.4
528.76	20.0	250.4	71.7
528.8 <sub>0</sub>	30.0	319.7	82.5
528.6 <sub>8</sub>	40.0	361.7	86.8
527.9 <sub>5</sub>	50.0	392.0	93.9
527.79	57.6	410.0	98.1
579.1 <sub>5</sub>	1.00	9.3	59.5
579.0 <sub>8</sub>	2.50	23.9	60.0
578.93	5.05	50.1	61.3
579.2 <sub>4</sub>	10.0	104.9	65.7
578.9 <sub>0</sub>	15.0	160.8	70.7
578.8 <sub>6</sub>	19.4	204.7	75.3
578.2 <sub>5</sub>	30.0	281.6	83.8
578.1 <sub>7</sub>	40.0	327.6	90.1
578.1 <sub>6</sub>	50.0	360.3	96.1
578.1 <sub>5</sub>	60.0	385.7	101.1
578.2,	70.6	407.5	105.9

Table AI. (Continued)

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